

Airman Knowledge Testing Supplement for Flight Instructor, Ground Instructor, and Sport Pilot Instructor

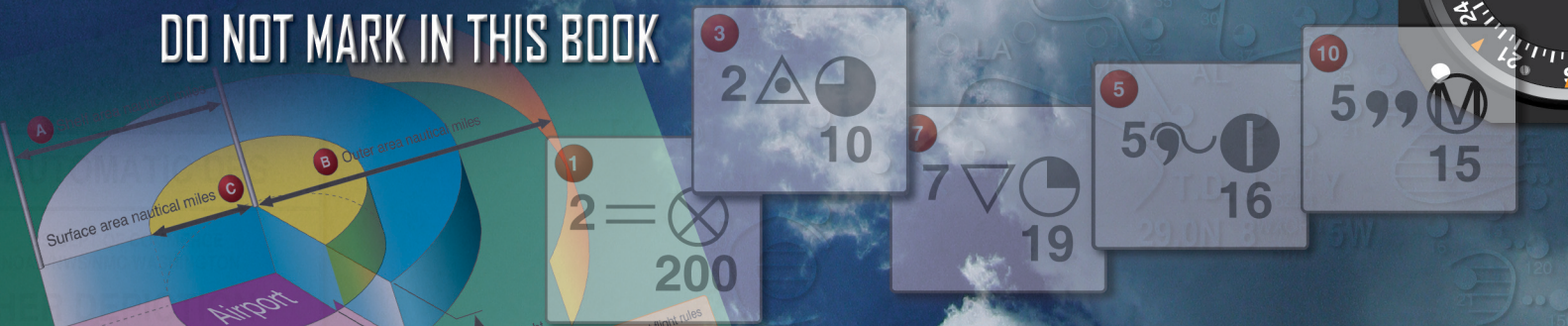


U.S. Department
of Transportation
**Federal Aviation
Administration**

Addendum A

*This publication was formerly known as:
"Computer Testing Supplement for Flight and Ground Instructor."*

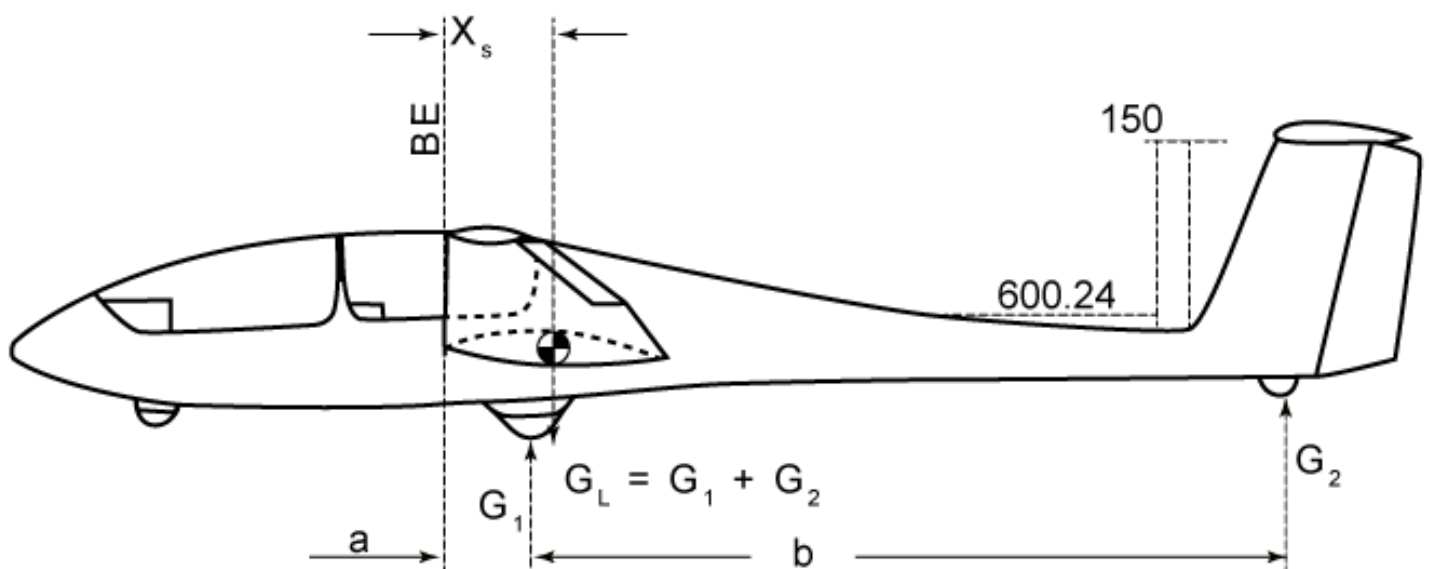
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Empty Weight		Range of C. of G. behind Datum			
kg	lbs	Forward		Aft	
		mm	inches	mm	inches
360	794	758	29.84	773	30.43
365	805	748	29.45	769	30.28
370	816	739	29.09	765	30.12
375	827	729	28.70	761	29.96
380	838	720	28.35	757	29.80
385	849	711	27.99	753	29.65
390	860	703	27.68	749	29.49
395	871	694	27.32	745	29.33
400	882	686	27.01	742	29.21

The weight of the non-lifting parts is the sum of the fuselage, tailplane, and maximum load in the fuselage and must not exceed 400 kgs (882 lbs). Otherwise, the maximum load permitted in the fuselage must be correspondingly decreased.



Datum Line: Front edge of the wing at the root rib.

Figure 57

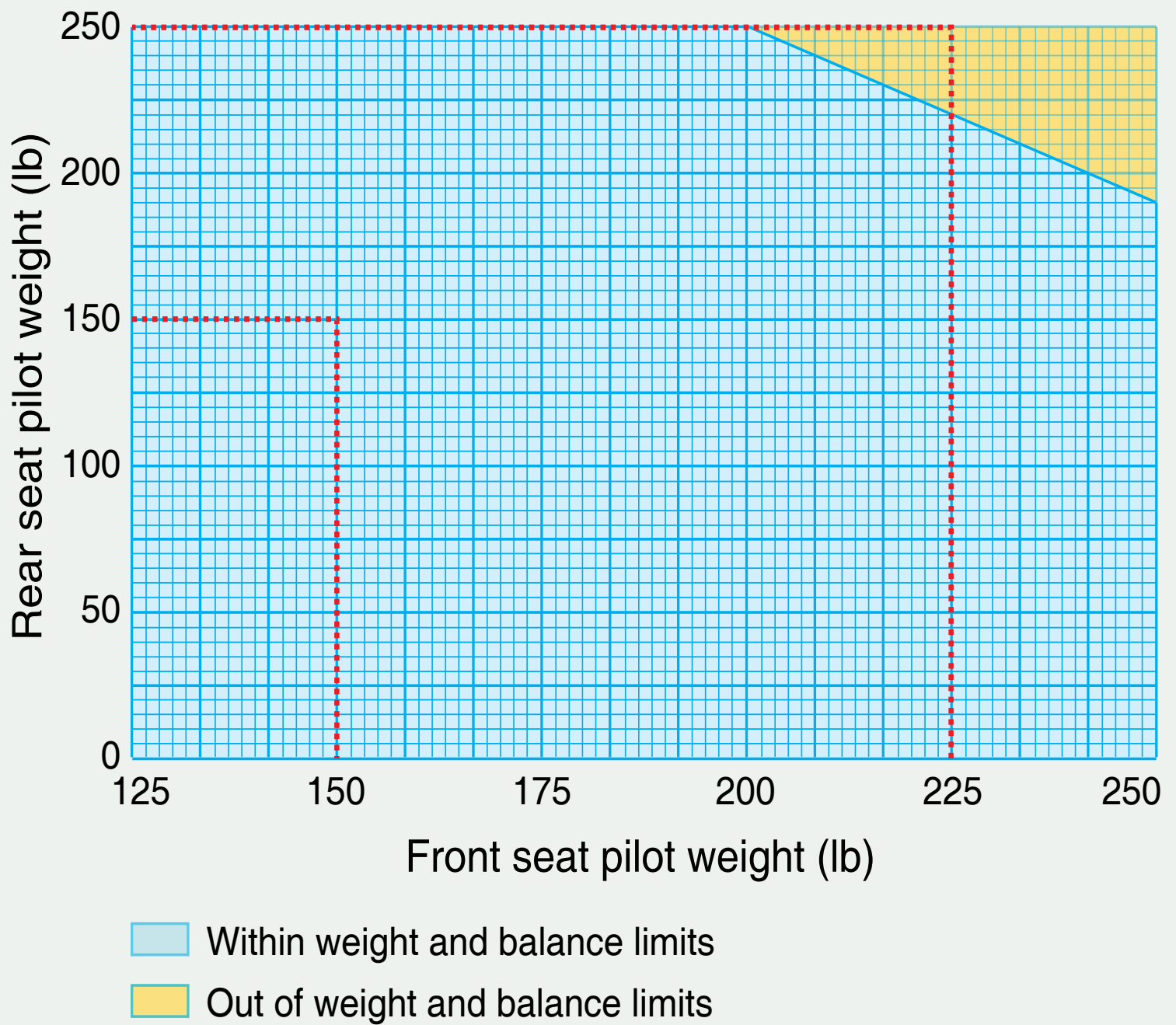


Figure 58